

# MECHANICAL-PROPERTY DATA Ti-10V-2Fe-3Al ALLOY

ISOTHERMALLY FÜRGED

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Prepared by

**BATTELLE** 

Columbus Laboratories Columbus, Ohio 43201

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### Ti-10V-2Fe-3Al Isothermally Forged

#### Material Description

Ti-10V-2Fe-3Al is a recently developed, metallurgically nearbeta alloy. The alloy is capable of attaining a variety of strength levels, depending on the selection of heat treatment. A major advantage over other alloys is the excellent forgeability. It forms readily at temperatures below those required for Ti-6Al-4V.

The Ti-10V-2Fe-3Al material used in this evaluation was received as 6 discs 7 inches (178 mm) in diameter x 1/2-inch (12.7 mm) thick. The material was produced by RMI and isothermally forged by TRV.

The chemical composition of this lct is as follows:

Chemical Composition	Percent Weight		
Vanadium	9.5		
Aluminum	3.2		
Iron	1.9		
Titanium	Balance		

## Processing and Heat Treating

A 30-inch (762 mm) diameter cast ingot was first heated to 2200 F (1478 K) and forged to a 24-inch (610 mm) round cornered square (RCS). The billet was then heated to 1400 F (1033 %) and forged to a 20-inch (508 mm) RCS, reheated to 1700 F (1200 K) and forged to a 15-inch (381 mm) RCS, reheated to 1700 F (1200 K) and once more forged to an 11-inch (279 mm) RCS bar. Conditioning of the piece was conducted as needed during the processing. A section of the material was then cut, heated to 1375 F (1019 K) and torged into an 8-inch (203 mm) RCS, reheated to 1700 F (1200 K) and forged to a 5-inch (127 mm) RCS, reheated to 1500 F (1089 K) and forged to a 4-inch (102 mm) octagon. A final pass at RMI was performed in a rotary forging machine at 1500 F (1089 K) transforming the octagon into a 3-1/4-inch (8.26 mm) diameter round bar.

At TRW the round bar was conventionally upset 25% at 1525 F (1103 K), conventionally drawn 40% at 1525 F (1103 K), isothermally drawn 50% at 1525 F (1103 K), and isothermally forged 50% to the final shape. The material was subsequently heat treated as follows: 1435 F (1043 K)/2 hours/air cool, 1425 F (1047 K)/2 hours/water quench, and 945 F (780 K)/8 hours/air cool (STA).

Condition: STA

Thickness: 1/2-inch (12.7 mm)

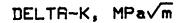
Properties	Temperature, F (K)			
	RT	(RT)	600	(587)
Tension				
TUS, L, ksi (MPa)	179.2	(1235.6)	153.2	(1056.3)
TUS, T, ksi (MPa)	178.3	(1232.8)	154.9	(1068.0)
TYS, L, ksi (MPa)	173.9	(1199.0)	136.6	(941.9)
TYS, T, ksi (MPa)	173.7	(1197.7)	141.2	(973.6)
e , L, percent in 2 in. (50.8 mm)	2.3		8.3 6.5 <sup>(b)</sup>	
e , T, percent in 2 in, (50.8 mm)	2.0		6.5 (b)	
E, L, $10^{\circ}_{3}$ ksi (GPa) $\binom{\text{d}}{\text{d}}$	15.46	(106.6)	13.98	(96.4)
e , T, percent in 2 in. (50.8 mm) E , L, 10 <sup>3</sup> ksi (GPa)(d) E , T, 10 <sup>3</sup> ksi (GPa)	15.12	(104.2)	13.98	(96.4)
Compression				
CUS, L, ksi (MPa) (b,c)	213.7	(1473.5)		
CUS, T, ksi (MPa)	211.4	(1457.6)		
CIIC T bed (MPa)	191.8	(1322.5)		
$E = I_{\rm o} \cdot 10^3 \text{ kg/s} \cdot (\text{GPa})^{(D, \Omega)}$	16.60	(114.4)		
E <sub>c</sub> , T, 10 <sup>3</sup> ksi (GPa) (d)	16.09	(110.9)		
Shear		<b>,</b> ,		
SUS, L, ksi (MPa)	98.6	(679.8)		
SUS, T, ksi (MPa)	101.5	(699.8)		
Bearing				
e/D = 1.5				
BUS, L, ksi (MPa)	247.1	(1703.8)		
BUS, T, ksi (MPa)	264.8	(1825.8)		
BYS, L, ksi (MPa)	243.0	(1675.5)		
BYS, T, ksi (MPa)	258.9	(1785.1)		
e/D = 2.0				
BUS, L, ksi (MPa)	321.2	(2214.7)		
BUS, T, ksi (MFa)	309.5	(2134.0)		
BYS, L, ksi (Mfa)	289.9	(1998.9)		
BYS, T, ksi (MPa)	280.8	(1936.1)		
Axial Fatigue (Transverse)				
Unnotched, R = 0.1				
10 <sup>3</sup> cycles, ksi (MFa)	170	(1172)		
10 <sup>5</sup> cycles, ksi (MPa)	120	(827)		
10 <sup>7</sup> cycles, ksi (MPa)	110	(758)		
Notched, $K_t = 3.0$ , $R = 0.1$ $10^3$ cycles, ksi (MPa)		4		
103 cyclēs, ksi (MPa)	105	(724)		
105 cycles, ksi (MPa)	45	(310)		
10 <sup>7</sup> cycles, ksi (MPa)	40	(276)		

<sup>(</sup>a) Values are average of triplicate tests conducted at Wright Aeronautical Laboratories, Materials Laboratory, unless otherwise indicated. Fatigue values are from curves generated using the results of a greater number of tests.

(c) Ultimate strength value only due to test problems.

<sup>(</sup>b) Data from only two tests.

<sup>(</sup>d) The values presented are indicative of modulus values typical for titanium alloy materials; however, the instrumentation did not meet ASTM E83 class A extensometer requirements.



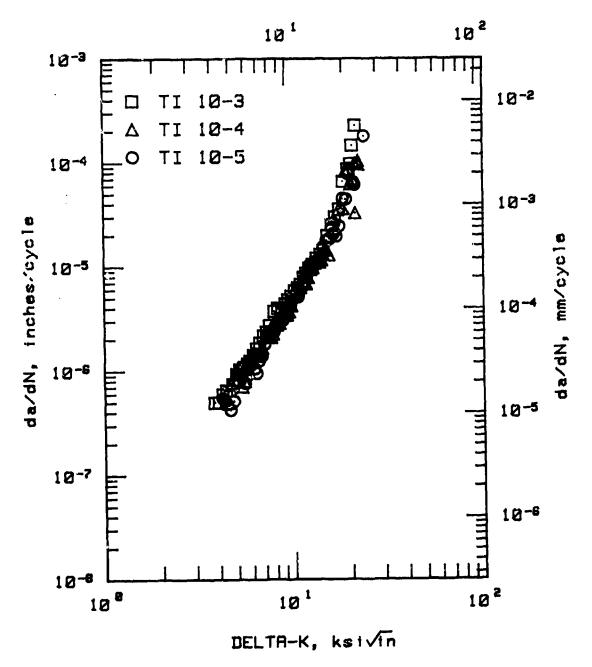


FIGURE 1. da/dN VERSUS DELTA K FOR Ti-10V-2Fe-3A1 ALLOY.

Lab Air
Room Temperature
B = 0.1
Frequency = 30 Hz
Specimen Orientation = L-T

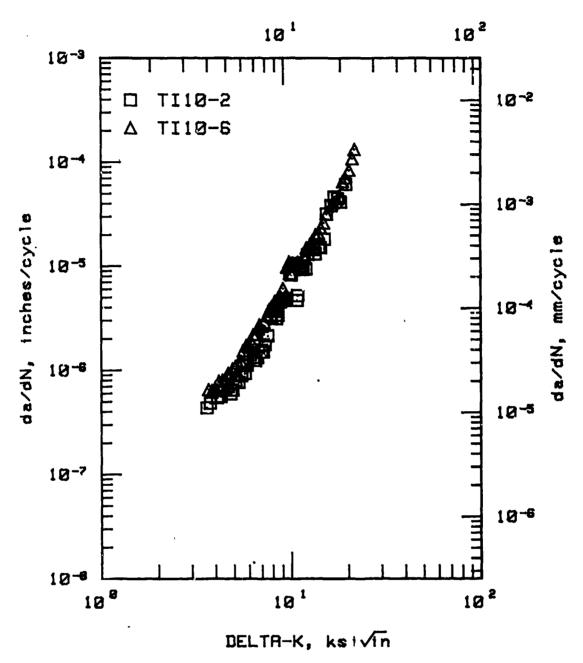
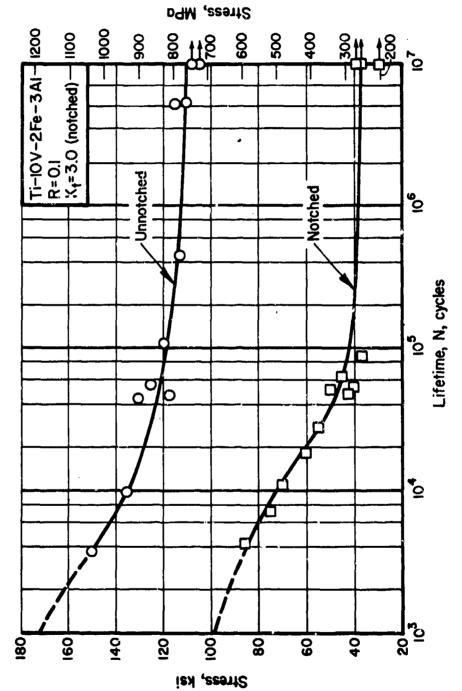


FIGURE 2. da/dN VERSUS DELTA K FOR Ti-10V-2Fe-3A1 ALLOY.

Lab Air (Heated) 600 F (589 K) R = 0.1 Frequency = 30 Hz Specimen Orientation = L-T



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AXIAL LOAD FATIGUE BEHAVIOR OF UNNOTCHED AND NOTCHED (K = 3.0) T1-10V-2Fe-3A1 ALLOY AT ROOM TEMPERATURE FIGURE 3.